

# NASA OBIGGS/OBOGS Research

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### Contents:

1. Where's NASA's plan going?
2. How will NASA do this?
3. NASA's in-house CO<sub>2</sub> OBIGGS work

# 1. What's with the OBIGGS / OBOGS?

OnBoard Inert Gas Generation System  
OnBoard Oxygen Generation System

## Why OBOGS

Task-sharing drives the cost down  
Oxygen required for high-altitude depressurization  
Pressurized oxygen system servicing needs  
highly-skilled personnel

## 1.1 Where does our work fit in?

### 2. Aviation Safety Program (Langley)

#### 2.5 Accident Mitigation (Glenn)

##### 2.5.1 Crash worthiness (Langley)

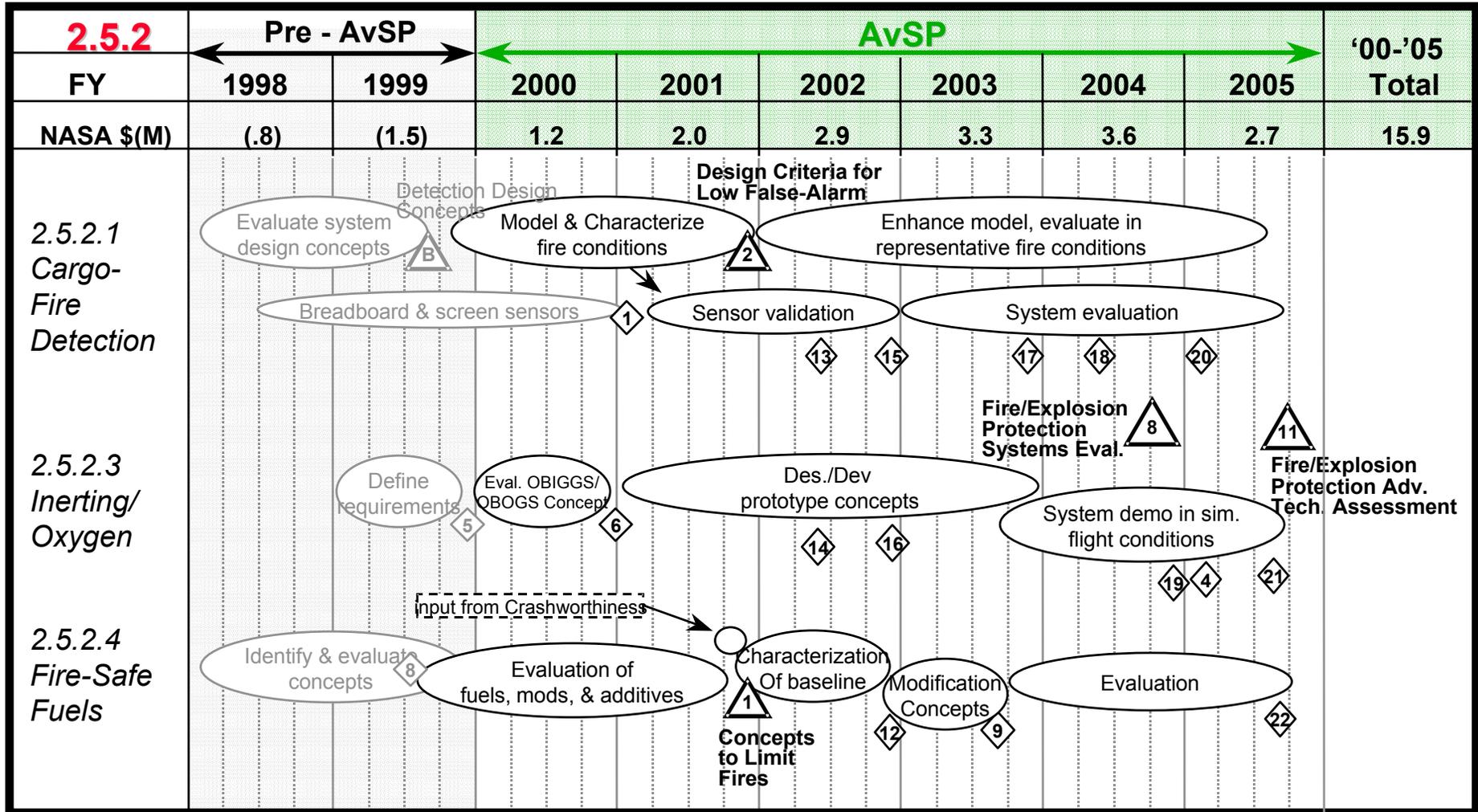
##### 2.5.2 Fire Protection (Glenn)

###### 2.5.2.1 Fire-Detection

###### 2.5.2.3 OBIGGS / OBOGS

###### 2.5.2.4 Fire-Safe Fuels

# 1.2 NASA Fire Protection Work Plan -- Milestones and budgets



### 1.3 What's our schedule?

<u>N<sub>2</sub>-based OBIGGS / OBOGS</u>		<u>Non-N<sub>2</sub> based OBIGGS /OBOGS</u>	
RFI & RFP to OEMs	11/01		
Eval. Phase I Proposals	01/02	Evaluate options	02/02
Award Phase I	02/02		
Complete Phase I	08/02	finish detailed design	08/02
<div style="border: 1px solid black; padding: 5px; display: inline-block;">Down-select</div>			
Award Phase II	10/02		
Complete Phase II	08/04	Finish hardware fabrication	08/04
Ground-based cargo fire-suppression test		12/04	
Integration into demo aircraft & safety clearance		12/04	
Flight test on B757		03/05	

## 2. Fostering Emerging OBIGGS / OBOGS Technology

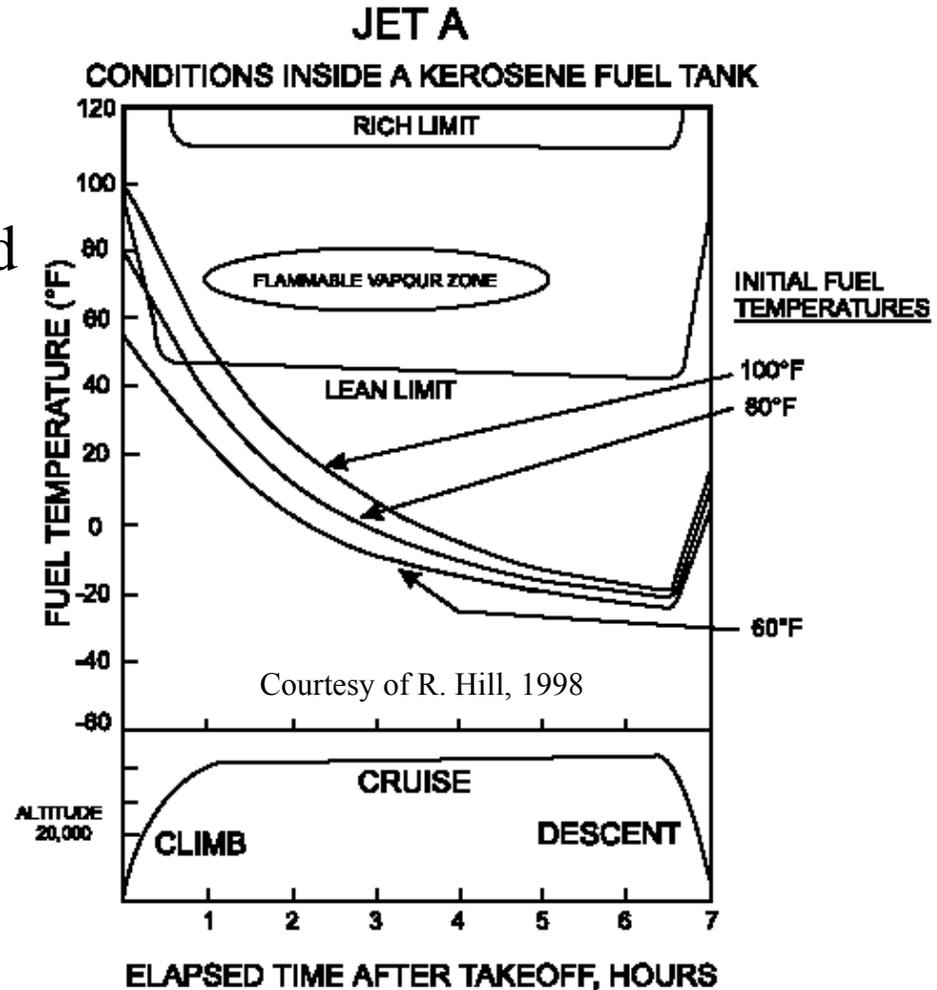
### 2.1 NASA-Sponsored Technology Demonstration on B757 in '04-'05

- NASA-Boeing relationship changed
- NASA will release the Request For Proposal directly
  
- Direct funding of OEM
  - Up to 6 Phase I feasibility studies for \$600K total in 6 months
  - Up to 3 Phase II hardware buildup, \$1.8M in 2 years total
  - May be more \$, depending on the technical nature of project
  
- NASA seed funding - OEM contribution needed
- “Technology development”, not “Prototype development”
  
- OEM retains ownership of hardware (under discussion)
- NASA OBIGGS/OBOGS funding:
  - FY02: ~1M, FY03: ~1.5, FY04: ~1.9M

## 2.2 Performance Requirements on B757-200 Flight Test

### 2.2.1 Center Wing Tank (CWT) inerting only

- Tank vented to atmosphere
- 10% O<sub>2</sub> concentration at departure time
- Assume 40 minutes turn-around between flights
- 7080 US gallons capacity
- Eliminate the fire hazard in first 1-1/2 hr
- Full-time purging during flight



## 2.2 Performance Requirements on B757 Flight Test

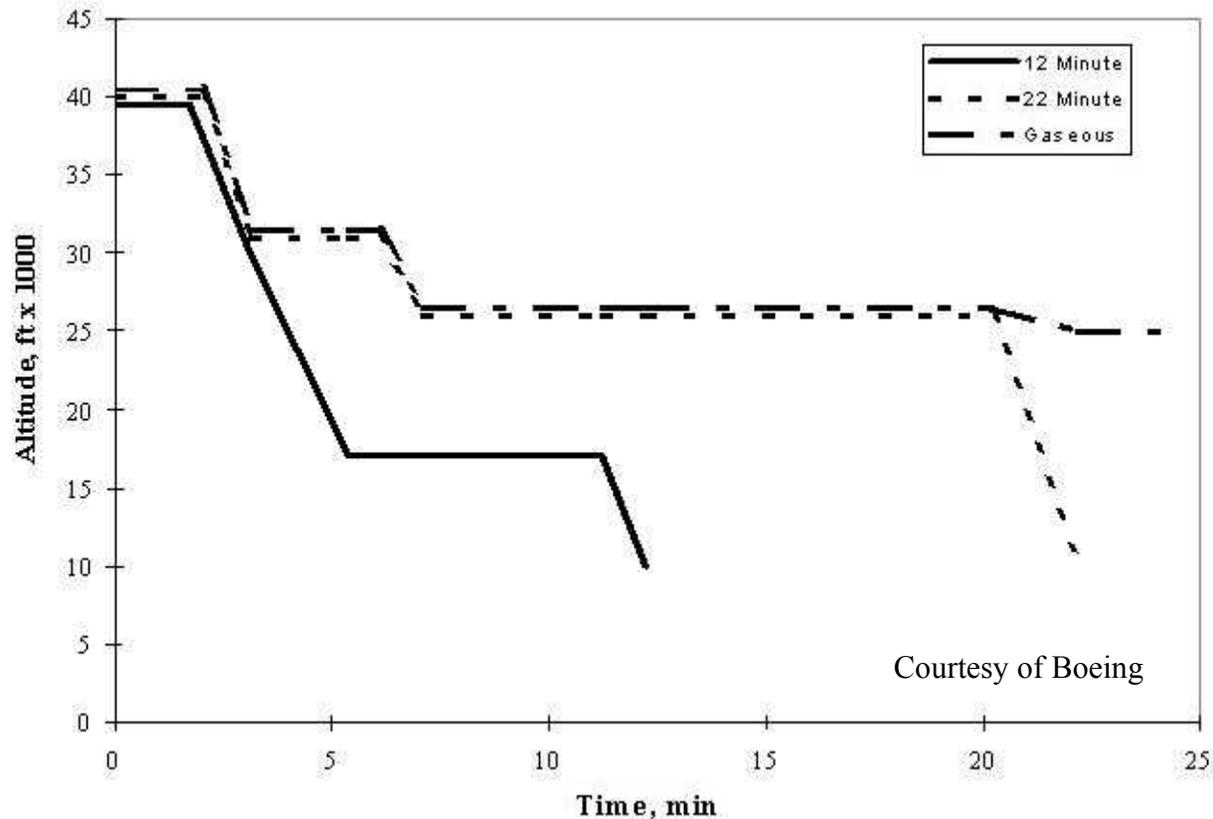
### 2.2.2 Cargo compartment fire suppression (tentative)

- 1300 ft<sup>3</sup> aft cargo compartment
- Assume initial knock-down already done
- Assume leakage of 11 ft<sup>3</sup> per minute
- Reach <12% O<sub>2</sub> concentration in 25 minutes  
at 8000 ft pressure and hold for 120 minutes
- OBIGGS not required to inert CWT when  
inerting cargo compartment

## 2.2 Performance Requirements on B757 Flight Test

### 2.2.3 Emergency OBOGS requirement (tentative)

- 35,000 ft depress.
- 10,000 ft in 12 min.
- 268 persons
- 93%+ pure O<sub>2</sub>
- ~2 liter O<sub>2</sub> (STP)  
/min/person  
or ~1.2 lbm/min  
peak flow



## 2.2 Performance Requirements on B757 Flight Test

### 2.2.4 Physical and resource constraints (tentative)

At the gate:

- No engine power
- APU provides ~35 lbm/min air at 35 psi
- But APU also needed to operate air-cond. and electrical sys.

Take off & cruise:

- Plenty of engine air and power
- Must conserve to reduce fuel-consumption

Descent & taxi:

- Little power and bleed air
- May get power and air if APU is used

Pallet-mounted, fit through normal 757 door, operated inside passenger cabin, minimal connections - power and bleed-air in, N<sub>2</sub> & O<sub>2</sub> out, cooling air in & out.

## 2.3 Where to get the information for the RFI and RFP

Commerce Daily

NASA GRC AvSP website:

[http://www.grc.nasa.gov/WWW/avsp/reports\\_fire\\_prevention.htm](http://www.grc.nasa.gov/WWW/avsp/reports_fire_prevention.htm)

Additional Boeing Reports

Task-1 Aircraft Requirements

Task-2 State of the Art OBIGGS / OBOGS

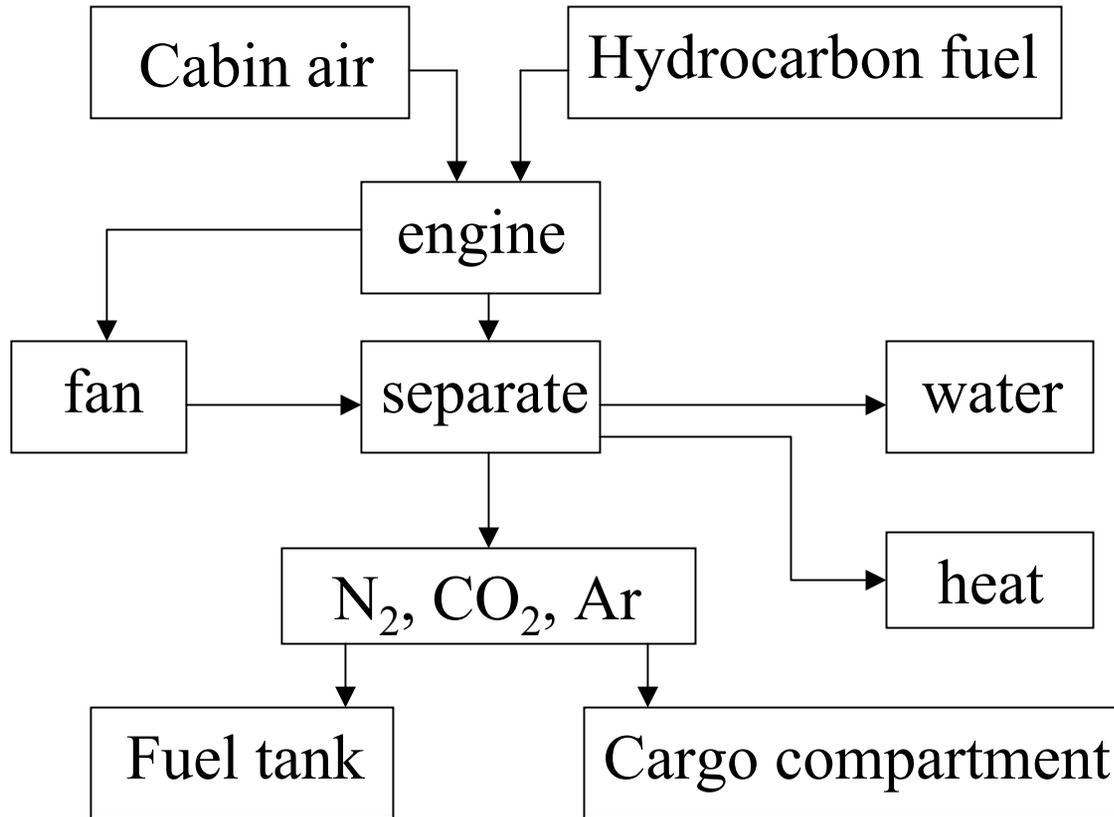
### 3. NASA's In-House OBIGGS Work

#### 3.1 Why a Non-N<sub>2</sub> based system?

- Fire doesn't care about what kind of inert gas.
- Air-separation is hard.
- Oxygen removal is easy.
- Burning air removes oxygen.
- You can make burnt air much faster than  
air separation

### 3. NASA's In-House OBIGGS Work

#### 3.2 Conceptual diagram



The devil is in the details, of course...

Here's our brick.  
Let's see your jade.

### 3. NASA's In-House OBIGGS Work

#### 3.3 How much fuel does it take to inert a B757 CWT?

species	fraction	mole count	wt	M	STP vol.	STP vol.	exhst fract
N2	0.782	66.4	1860	28			0.84
O2	0.209	17.8	568	32			
Ar	0.009	0.8	31	40			0.01
total air		84.9	2458	28.94	2.076		
C		12	144	12			
H		23	23	1			
total fuel			167			0.055	
CO2		12	528	44			0.15
H2O		11.5	207	18			
dry total		79.2	2418	30.5	1.936		
units		mole	gm	gm/mol	m <sup>3</sup>	gal	

B757 CWT: 7080 gal = 27 m<sup>3</sup> => 0.76 gal of fuel

#### How long does it take to fill up B757 CWT?

Assume a 5 HP lawn mower engine with 25%  $\eta$ ,

$$\Delta h_{\text{fuel}} = 42 \text{ MJ/kg}, \rho_{\text{fuel}} = 6.7 \text{ lbm/gal}$$

it consumes fuel at 1.3 kg/hr or 0.4 gal/hr

or [O<sub>2</sub>] from 20% to 10% in ~o(1 hr.)

### 3. NASA's In-House OBIGGS Work

#### 3.4 Other CO<sub>2</sub> issues under investigations

- How will dissolved CO<sub>2</sub> affect pump cavitation?
- How much more ignition delay will CO<sub>2</sub> chemical equilibrium introduce?
- How much will dissolved CO<sub>2</sub> retard fuel coking?
- Will CO<sub>2</sub> blanketing the fuel and retard fuel vaporization?

#### 4. Summary

NASA intends to develop OBIGGS/ OBOGS to offset the cost of inerting fuel tanks alone.

NASA intends to fund development of emerging technologies for ground-based technology demonstration in FY04 and flight test aboard the B757 (type of A/C) in FY05.

RFP for multiple-award \$70k Phase I seed funding for 6-month feasibility study will be announced in 11/01. Check with NASA Glenn Aviation Safety web-page for details.

NASA's internal CO<sub>2</sub> program suggests relatively simple and feasible means of inerting fuel tank and suppressing cargo compartment fire.